



Wallops Flight Facility (WFF) Environmental, Integration and Test, and Fabrication Capabilities

Wallops Environmental, Integration, & Testing Capabilities





Goddard Space Flight Center

Electro-Mechanical Fabrication Facility



Telemetry Ground Stations

Vibration

Test



Payload Integration Labs



Spin Balance



100K Clean Rooms & 10K Clean Tents



EMI/RFI Test Chamber



Thermal Vacuum Chamber





Bend Testing



Wallops Environmental, Integration, & Testing Capabilities





Magnetic Calibration Facility



Moment-Of-Inertia Testing



Antenna Patterning Chamber



Spin Deployment Facility



Attitude Control System Test Lab



Mission Planning Lab



GPS Simulation & Test Lab



Thin Film Material Test Lab



Wallops Island Environmental, Integration & Testing Capabilities

Goddard Space Flight Center





WFF

- *MIL-STD 461/462 Compliant and ANSI C63 Pre-Compliant Shielded EMI/EMC Measurement Facility for Radiated and Conducted Susceptibility and Emissions measurements.*
- Available Tests: CE102, RE102, and RS103
- 12 ft. Wide x 18 ft. Long Main Chamber is lined with an absorber rated up to 40GHz.
- Shielded (100 dB to 40 GHz) Chamber, Control Room, and Payload Support Equip. Room
- The chamber also provides a low noise environment that could allow for RF sensitive device characterization.





Bldg F-7 Antenna Pattern Measurement Facility

- A shielded Dual-Mode Far Field/Compact Range Chamber for measurement and analysis of antenna pattern data.
 - Far-Field mode: 400 MHz 2 GHz frequency range.
 - Compact Range Mode: 2 GHz – 100+ GHz frequency range. 6'W x 6'D x 4'H elliptic cylinder quiet zone
- The Agilent PNA Orbit/FR Spectrum based data collection system allows both Phase and Amplitude (Linear & CP) Pattern Measurements with data export to Microsoft Excel.
- Maintain feeds and standard gain horns for UHF, L-Band, Sband, C-band, X-band, Ku-Band, and Ka-Band.
- Chamber Shielded 90 dB to 40 GHz





Bldg F-7 Thermal Vacuum Chamber



Space-rated Thermal Vacuum Chamber 4 ft x 6 ft horizontal chamber **Space Qualification Testing** 1 X 10-6 Torr **Balloon and aircraft testing** 0.25 to 760 torr Vacuum System: Oil-free mechanical pump and cryogenic pump **Thermal Simulation:** Twelve independently controlled stainless steel panels **Contamination Monitoring:** TQCM, cold finger, scavenger plate Instrumentation:

Type-T TCs and a Fluke 2686A DAQ with PAI modules

Usable shroud envelope:

4' diameter

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5' 11" length

Full opening door, bracket hinged. Six instrumentation/power feedthrough Ports.







Bldg F-7 High Bay East



- Class 100,000 cleanroom with an ESD floor.
- 40 ft. x 60 ft. cleanroom with 20 ft. x 40 ft. Airlock/Ante Room
- 5-Ton Bridge Crane; 25 ft. hook height; Remot Controls











9/20/12







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Capabilities

Usable Envelope	70" wide x 21" deep x 46" tall interior Stainless Steel tabletop
Flow	HEPA filter 99.99% efficient Velocity of approximately 90 FPM Designed to provide an ISO Class 5 (Class 100) or ISO Class 4 (Class 10) clean air environment
Electrical	Motor speed is variable with Solid State controller Electrical outlet is a 15 amp duplex installed in the base right front under the table System provides independent motor and light on/of switches
Motor/Blower	System utilizes a PSC direct drive motor with a dynamically balanced blower wheel Assembly is double vibration isolated from cabinet to provide excellent vibration control





Bldg E109 Vibration Test System

Capabilties

Fixture	10" X 10" high strength magnesium alloy mounting fixture
Shaker	Sine and Random Rating: 2200lbs Maximum Shock: 4500lbs Frequency Range: 5Hz-2kHz
Accelerometers	Various sizes Single axis Tri-axis
Software	Unholtz-Dickie Corporation
Performance	Sine/Random/Shock

Applications:

- Space qualification testing
- Environmental testing for electronic flight hardware

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Bldg E109 GPS Simulator Lab

- 4 Spirent simulators capable of up to 4 RF outputs of GPS L1/L2/L5
- Able to model a range of antenna patterns
- Simulation capability able to simulation a range of trajectories along with simulated IMU output for HWIL and SWIL testing



Autonomous scenario run capability for Monte Carlo, parametric run testing for large number of HWIL/SWIL tests

Bldg E109 Solar Simulator/TVac Chamber



E109 Solar Simulator

- 3 kW Fresnel Lens, Xenon Lamp Solar Simulator
- Class ABA*

Goddard Space Flight Cen<u>ter</u>

- 11" Beam diameter
- ~1350 W/m² max output
- Two-axis gimbal and linear
 actuators available
- E109 TVac Chamber
- Pressure as low as 0.1 Torr
- Temperature: -70^c to +170^c
- Dimensions: 21"D"x24"H"x21"W"







E109 Solar Simulator

E109 TVac



Bldg E109 Radar Lab Facility



- Radar Transponder Certification Bench.
- Flight Termination Receiver System (FTS) Certification Bench
- Test Equity Model 115 Thermal Cycle Chamber





Flight Termination Receiver System (FTS) Certification bench

Note Flight Center Bldg E109 Microwave Lab Facility

- Cortex XXL ™ Telemetry Receiver System
- Agilent/Keysight
 Microwave Network and
 Spectrum Analyzers
- Microwave Telemetry LCT2 Transceiver Software Designed Radio Development and Test Area
- Field Programmable Gate Array (FPGA)
 Design and Test Area 9/20/12









E-109 Mission Planning Lab (MPL) – SmallSat/CubeSat Design





		Cycle I	Cycle Length Science Mo		de (Deploy) Science Mode ((Deploy) + Tx Detumble M		e Mode
Subsystem	Component	Power (W)	(hours)	Duty Cycle	WHrs Used/Cycle	Duty Cycle	WHrs Used/Cycle	Duty Cycle	WHrs Used/Orbit
	CMOS	0.6	1.5	0.22200	0.200	0.22200	0.200	0.000	0.000
Experiment	Interface Board	1	1.5	0.22200	0.333	0.22200	0.333	0.000	0.000
	Filter Motor	5	1.5	0.00556	0.042	0.00556	0.042	0.000	0.000
	Blue Canyon XACT	1.2	1.5	1.00000	1.800	1.00000	1.800	1.000	1.800
	Fine Sun Sensor	1	1.5	1.00000	1.500	1.00000	1.500	1.000	1.500
	Thrusters	0.3	1.5	1.00000	0.450	1.00000	0.450	1.000	0.450
	Photo Diodes	0.5	1.5	1.00000	0.750	1.00000	0.750	1.000	0.750
	Tx	6.4	1.5	0.00000	0.000	2.67000	25.632	0.000	0.000
Comm	Rx	6.4	1.5	1.00000	9.600	1.00000	9.600	1.000	9.600
	Antenna	0.0264	1.5	1.00000	0.040	1.00000	0.040	1.000	0.040
C&DH	CHREC	2.58	1.5	1.00000	3.870	1.00000	3.870	1.000	3.870
Devuer	EPS	0.1	1.5	1.00000	0.150	1.00000	0.150	1.000	0.150
Power	Custom Card	1	1.5	1.00000	1.500	1.00000	1.500	1.000	1.500
Batteries	Consumption	0.3	1.5	1.00000	0.450	1.00000	0.450	1.000	0.450
Heaters	Batteries	0.594	1.5	0.10000	0.089	0.10000	0.089	0.100	0.089
Total Consumed		27.0004			20.773		46.405		20.199
Consumption EPS L	osses =	10.0%			2.077	Ī	4.641		2.020
Consumption Harn	ess Losses =	5%			1.039		2.320		1.010
Consumption Batte	ery Losses =	5%			1.039		2.320		1.010
Total Consumed (In	nc EPS, Harness and I	Battery Losses)			24.928		55.686		24.238
Worst Case Enegy Production (Beta = 0, Orbit Average X Orbit Length)			56.3		56.3		0		
Charging EPS Losses = 15.0%				8.445		8.445		0.000	
Charging Battery Losses = 5.0%			2.815		2.815		0.000		
Worst Case Enegy Production (Including EPS and Battery Losses)			45.040		45.040		0.000		
Net Whrs Per Cycle (Production - Consumption)				20.112		-10.646		-24.238	
# orbits on a typical day (Orbit length = 90 minutes)			15.00		1				
Net Whrs on a day (Net Whrs per Orbit * #Orbits)				301.682		-10.646			











Bldg F10 Sounding Rockets Facilities



Bldg F10 Vibration Testing

- There are four Ling Electronics shakers used for component and payload vibration tests at WFF with the following technical information.
- NSROC Shaker Specifications
- Ling Electronics Shaker B340
 - Rated Force Sine: 30,000 lb
 - Rated Force Random:30,000 lb rms
 - Frequency Range: 5-2000 Hz
 - Maximum Displacement Peak-Peak: 1 inch
 - The B340 can be rotated to mate with a TEAM Corp. model 482 sliding table so that it can be used for both thrust axis and lateral vibration tests. Max. pitch moment capacity = 1,200,000 lb.-in.
- Ling Electronics Shaker B335 (x2)
 - Rated Force Sine: 18,000 lb
 - Rated Force Random:18,000 lb rms



Bldg F10 Vibration Testing(cont)



• Ling Electronics Shaker B335 (x2, continued)

- Frequency Range: 5-3000 Hz
- Maximum Displacement Peak-Peak: 1 inch
- The B335 can be rotated to mate with a TEAM Corp. model 482 sliding table so that it can be used for both thrust axis and lateral vibration tests. Max. pitch moment capacity = 240,000 lb.-in.

Ling Electronics Shaker B395

- Rated Force Sine: 6000 lb
- Rated Force Random:5,750 lb rms
- Frequency Range: 5-3000 Hz
- Maximum Displacement Peak-Peak: 1 inch
- The WFF Shaker test facilities are equipped with 11 in. cube fixtures so that tests can be performed on small components in all three axes by mounting the test article in different orientations. At the engineer's or Principal Investigator's request, sensors can be mounted on any part of the payload to monitor its response. WFF Shaker test facilities have 16channel capability

Bldg F10 Static and Dynamic Balancing Machines

- At WFF, a Gisholt Rocket Balancing Machine is used to balance sounding rocket payloads (Figure
- 6.2.2-1 This machine's specifications are listed below.
- • Max. payload weight = 1500 lb.
- Max. height of CG above table = 10 ft.
- Measurement accuracy = 2.0 ozin2 at 225 rpm or more



Bldg F10 Airdyne Mark 8 Mass Properties Measurement System

- The Environmental Testing and Evaluation Group at WFF is equipped with an Airdyne Mark 8 mass properties measurement system. This unit is used for measuring center of gravity (CG) locations and moments of inertia (MOI) on sounding rocket subsystems and payload stacks. Important technical data include:
- Maximum test article weight: 5,000 lb.
- • Maximum CG height above the table: 120 in.
- CG and MOI measurement accuracy: 0.1%









Attitude Control System Test Lab

Bldg F10 Spin Deployment Facility

6.2.6 Spin Deployment and Separation Equipment (WFF)

Payloads with deployable booms, nose cones, doors, etc. can be tested for proper operation using the spin deployment and separation chamber at WFF. The rotary table is capable of spinning a payload to a rate of 20 rps while withstanding an imbalance of up to 3000 ft-lb. 5 ft. above the table surface. The chamber is equipped with a heavy-duty Kevlar[®] tarp around the rotary table for catching deployed components. Also, there are video cameras mounted on the chamber walls for recording and timing the deployment events. Pyrotechnic release devices can be activated by connecting lead wires through a 20-channel slip ring that allows the table to rotate while maintaining electrical continuity.



Spin Deployment Facility

Bldg F10 Bend Test Facility

6.2.5 Bend Test Fixtures (WFF and WSMR) Every sounding rocket payload is subjected to a bend test in order to determine the overall stiffness of the body. This information is used by the Flight Performance Group to verify payload stability during flight. The bend test fixtures at WFF and WSMR consist of a base plate mounted to the concrete floor and a pneumatic (WFF) or motor driven (WSMR) linear actuator mounted to a steel I-beam pillar. The pistons are equipped with load cells, which are used to measure and control the applied load. The aft end of the payload is fastened to a base plate, and the actuator's position along the pillar can be adjusted to the proper height on the payload being tested. Land surveying equipment is used to accurately measure the tip deflection of the payload as the actuator applies lateral loads in both directions.



Figure 6.2.5-1: Bend Test Fixture at WFF

Facility	Maximum Load (actuator or load cell)	Maximum Actuator Height	Accuracy of Deflection Readings
WFF	+/- 5,000 pounds	21 feet	0.05 inches
WSMR	+/- 1,150 pounds.	21 feet	0.05 inches

Table 6.2.5-1 NSROC Bend Test Fixture Specifications



Bldg F10 Centrifuge Test Facility

6.2.7 Centrifuge Machine (WFF)

A Genisco Model 1068-2 centrifuge machine is used for component acceleration tests at WFF. It is capable of achieving up to 1000 g acceleration at a radius of 10.5 in. It has 8" of clearance between the 3' diameter rotary table and the cover.



Figure 6.2.7-1 Genisco Centrifuge Machine at WFF



Bldg F10 Thermal Vac





Figure 6.2.4-1 PV/T Vacuum Chamber (left) and Tenney Space Simulation System Thermal Vacuum Chamber at WFF

Manufacturer	PV/T Inc.	Tenney Space Simulation System	Tenney Space Jr.
Inside Dimensions (ft. dia. x ft. lg.)	7x12	2x2	1.2x1.0
Minimum Pressure (torr)	2 x 10 ⁻⁵	3 x 10 ⁻⁸	7.5 x 10 ⁻⁸
Temperature Range (°C)	N/A Heat lamps used if needed	-73 to +125	N/A

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WFF Magnetic Test Facility

6.2.8 Magnetic Test Facility (WFF & WSMR)

At WFF, this facility is used to conduct magnetic calibration of magnetometers on sounding rocket payloads and to perform functional tests on magnetic attitude control systems. When required, magnetic calibration tests are done - generally for all payloads with magnetometers except those in which the magnetometers are used as roll or yaw indicators. The testing equipment consists of a three axis, 40 ft. square Braunbek system which is capable of canceling the effects of the earth's magnetic field and then generating a test field in any direction. Technical data are listed below.



Figure 6.2.8-1 The Magnetic Test Facility at Wallops

WFF Magnetic Test Facility(cont)

NASA

- Resolution = 10 nanotesla
- Field magnitude = 0 to 65,000 gamma.

Table 6.2.8-1: Instrumentation Available at the WFF Magnetic Test Facility

Item	Function	Specifications	Model
Proton Magnetometer	Calibration	Range: 20K-120K Gamma	GEM
		Resolution: 0.01 Gamma System	
		Accuracy: 0.2 Gamma GSM-19	
Triaxial Fluxgate	Test	Range: <u>+</u> 100K Gamma	EMDS
Magnetometer	Instrumentation	Resolution: 3 Gamma	SDM-313
		Orthogonality: 25 Arcmin	
Triaxial Fluxgate	Ambient Sensor	Range: <u>+</u> 100K Gamma	EMDS
Magnetometer	(outside)0	Resolution: 3 Gamma	SDM-313
		Orthogonality: 25 Arcmin	
Payload Magnetometer	Test	Range: <u>+</u> 100K Gamma	Bartington
	Instrumentation	Resolution: 3 Gamma	Mag-03MRN
		Orthogonality: 1 Degree	
Theodolites	Alignment	Resolution: 20 Arcsec Dicarlo	Theo020B
RF Horn Antenna	Data Receiving Freq.	Range: 1-18 GHz	Emco
		Gain: 7 dB Model 3115	
RF to Fiber-Optic	Data Conversion Freq.	Range: .1-5 GHz	Ortel
Transmitter	_	Watts: 6.4 mW 3450A-20	

WFF Magnetic Test Facility(cont)





Table 6.2.8-2: Magnetic Test Facility Specifications

Physical Dimensions: Access Opening Static Field Environment: Magnitude (each axis) Step Resolution Stability

Homogeneity Dynamic Field Environment: Magnitude Frequency

Turntable Coil Orthogonality Fields

8'8" H x 7'5" W

±100K Gamma
±3.7 Gamma
±10 Gamma/minute for first 30 minutes
±3 Gamma/minute after 60 minutes
0.02%, 6 ft. spherical diameter

+60K Gamma 10 Hz, Although 10 Hz to 100 Hz @ 1K Gamma has been performed 4' Diameter 1.8 Arcmin, Calibrated on 9/27/96 Earth, 0-15 Volts DC, 0-25 Amps Test, (3-Axis) 50 Volts AC, <u>+8</u> Amps Gradient, 15 Volts DC, 6 Amps

Bldg F10 Payload Fabrication





Bldg F10 Payload Fabrication

- NASA
- Wallops has a fully equipped machine shop that can provide electronic, electrical and mechanical support. The 26,000-square-foot machine shop includes a large selection of Computer Numerically Controlled (CNC) mills and lathes, manual machines, sheet metal fabrication, welding and heat-treating facilities. Capabilities include full CAD/CAM implementation in developing and fabricating mechanical systems, optical instrumentation, and payload components for flight research. The fabrication area performs functions such as sounding rocket launcher refurbishment, design and fabrication of mobile telemetry and mobile radar support vans and antenna systems. The machine shop includes mechanical technician laboratories for assembly of scientific sounding rocket payloads. While the facility primarily supports the Sounding Rocket Program, it regularly supports other NASA and reimbursable projects. The facilities are managed through the NASA Sounding Rocket Operations Contract (NSROC).

NASA

Bldg F10 Payload Integration & Testing

- Bldg F10 has four Independent Payload Integration and Testing (I&T) Areas
- Each I&T Area has its own Independent Telemetry Ground Station





Bldg F10 Telemetry Ground Station (1 of 4)





Frequencies: Modulation Formats: # of Channels:

Wallops Island Environmental, Integration, & Testing Capabilities







Wallops Island Assembly and Payload Processing Facilities

Fable 3-1. Assembly and Payload Processing Facilities Wallops Island						
Building	Function	Sq.Ft.	Special Features			
V-45	Assembly	4,933	• 10-ton bridge crane – critical			
V-55	Assembly	2495	• 20-ton bridge crane – critical			
W-15	Assembly	5,165	 one 3,936 sq. ft. bay door 13 ft high x 12 ft wide 3-ton overhead crane with 10-ft hook height approved for explosives 6-ton bridge crane – non-critical 			
W-40	Assembly	5,255	• 6-ton bridge crane (dual 3-ton trolleys) – non-critical			
W-65	Assembly	13,255	• 6 bays• pyrotechnic storage rooms• 6 assembly bays• approved for explosivesBay DoorsHxWCrane(s) hook height (hh)Bay 17 ft 10 in x 23 ft 11 in2x10 ton bridge/20 ft hhBay 218 ft x 23 ft 11 in2x7.5 ton monorail/18 ft hh17 ft 10 in x 23 ft 11 in2x3 ton monorail/18 ft hhBay 317 ft 10 in x 18 ft 11 inBay 414 ft 11 in x 15 ft 11 inBay 514 ft 11 in x 15 ft 11 inBay 614 ft 11 in x 23 ft 11 in			

Wallops Island Assembly and Payload Processing Facilities (cont)

Building	Function	Sq.Ft.	Special Features			
X-15	Payload processing	5,740	 co-located optical and crash/fire/rescue facilities door 19 ft 10 in high and 18 ft 10 in wide 3-ton overhead crane with 19-ft hook height laboratory and office space 1-ton stationary electric chain hoist – non-critical 1-ton electric chain hoist – non-critical 5-ton bridge crane – non-critical 			
Y-15	Assembly	8,240	 one high seven oth approved Bay Doors Bay 1 Bay 2 Bay 2 Bay 3 Bay 4 Bay 5 Bay 6 Bay 7 Bay 8 	bay (Bay 8) er bays for explosives HxW 9 ft 6 in x 17 ft 6 in 6 ft 10 in x 8 ft 6 ft 10 in x 8 ft 10 in x 8 ft 13 ft 7 in x 10 ft 10 in	Crane(s) hook height (hh) 3-ton monorail/7 ft 10 in hh 3-ton monorail/7 ft 10 in hh 2-ton bridge/15 ft 10 in hh	

Wallops Fabrication, Integration, & Testing Capabilities





Figure 3-10. H-100 Payload Processing Facility (PPF)

Figure 3-11. V-55 Spacecraft Fueling Facility (SFF)